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# High Thermal Conductivity NARloy-Z-Diamond Composite for Advanced Rocket Engines

### **Objectives:**

- Develop high thermal conductivity NARloy-Z-Diamond composite (NARloy-Z-D)) material for advanced rocket engines
- Develop near net shape fabrication technique for NARloy-Z-D combustion chamber liner

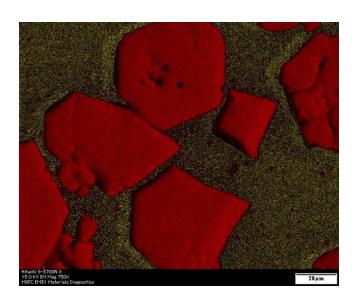
#### **Technical Goals:**

- Significant improvement in thermal conductivity over state of the art NARloy-Z alloy
  - -- up to 2X
- Significant improvement in the performance of combustion chamber liner made from NARloy-Z-D

## NARloy-Z-40vol.% Diamond Composite Microstructure Z-D

### **Target Applications:**

- Combustion chamber liner for advanced rocket engines
- Thermal management systems for nuclear propulsion system



#### **Research Team:**

**NASA Principal Investigator:** 

Dr. Biliyar N. Bhat/Marshall Space Flight Center (MSFC/EM31)

**NASA Co-Investigator:** 

Dr. Sandra Greene (MSFC/ER32)

**External Co-Investigator:** 

Dr. Jogender Singh/Pennsylvania State University (PSU) – Applied Research Laboratory (ARL)

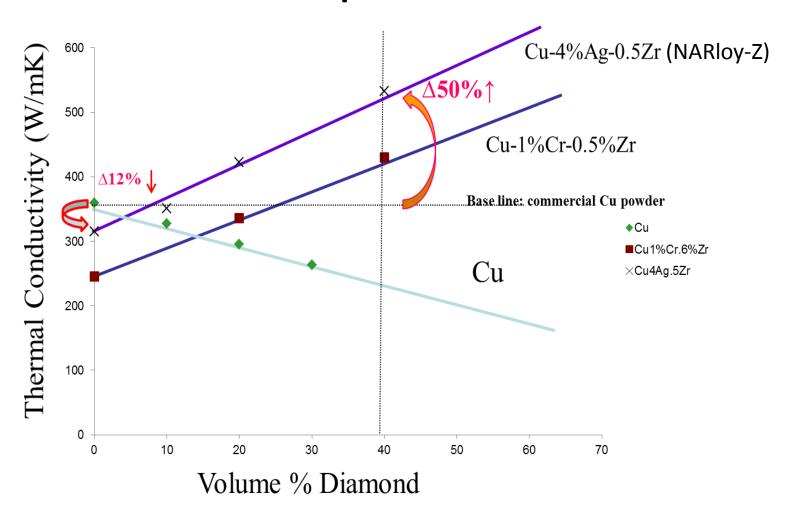
**Consultant:** 

Dr. David Ellis (NASA-GRC)

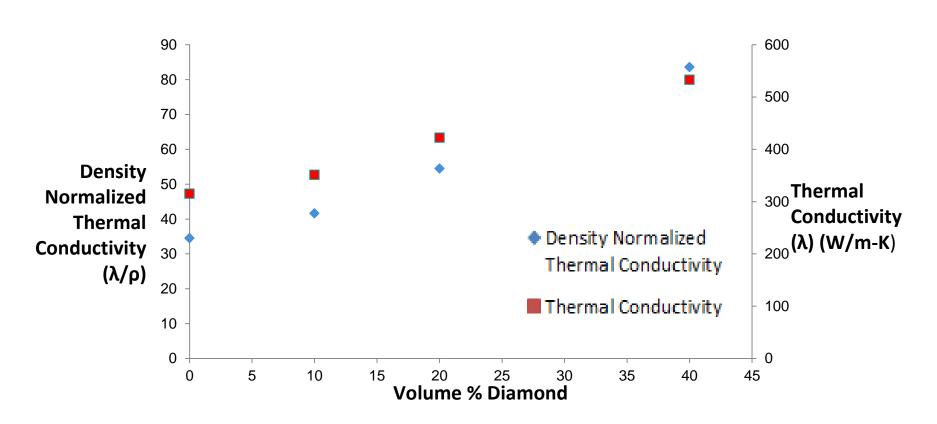
### **Technical Approach**

- Blend NARloy-Z powder with up to 40 volume% of diamond powder to produce NARloy-Z-D composite mixture
- Sinter at elevated temperature using Field Assisted Sintering Technology (FAST) at Pennsylvania State University (PSU)
- Characterize NARloy-Z-D composite
  - Microstructure analysis: SEM, TEM, XPS, EDS
- Develop design properties
  - Thermal conductivity measurements RT to 1000°F
  - Tensile testing at room and elevated temperatures
- Net shape forming of combustion chamber liner using FAST
- Hot fire testing of combustion chamber liner

# Thermal conductivity of NARloy-Z-Diamond Composites

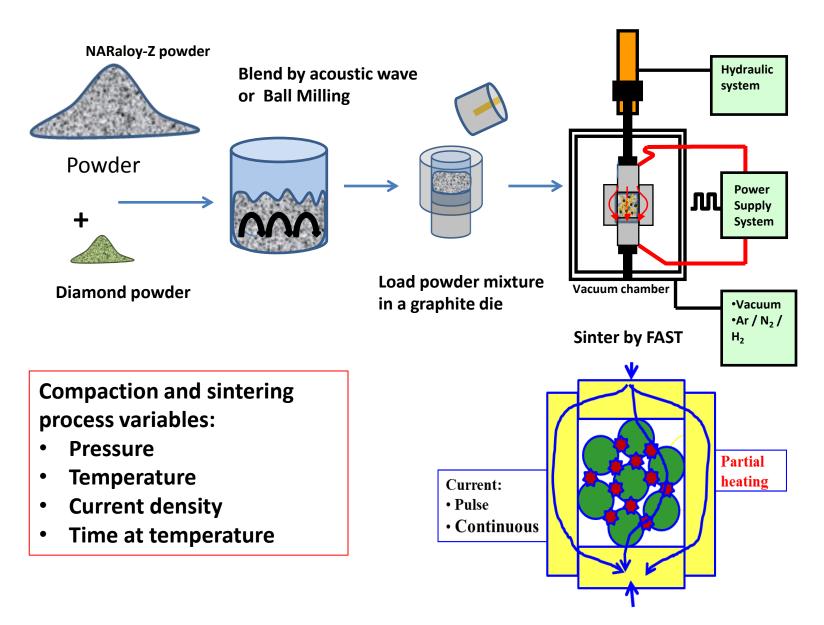


# Density Normalized Thermal Conductivity of NARloy-Z-Diamond Composites



140% improvement in density normalized thermal conductivity at 40 Vol.% Diamond

# Sequence of powder mixing and sintering by Field Assisted Sintering Technique (FAST)



### Field Assisted Sintering System At Penn State - ARL



• 250 ton Prototype Large R&D system

• Maximum Diameter: 300 mm

• Pulse current: 0-10KAmps

• Pulse time: 1 to 1000 ms

• Pause duration: 0 to 1000 ms

• Temperature capability: RT to 2400 °C

• Computerized Process control system

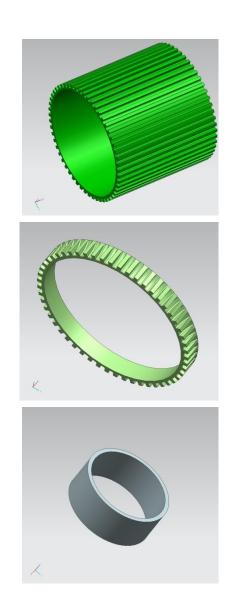


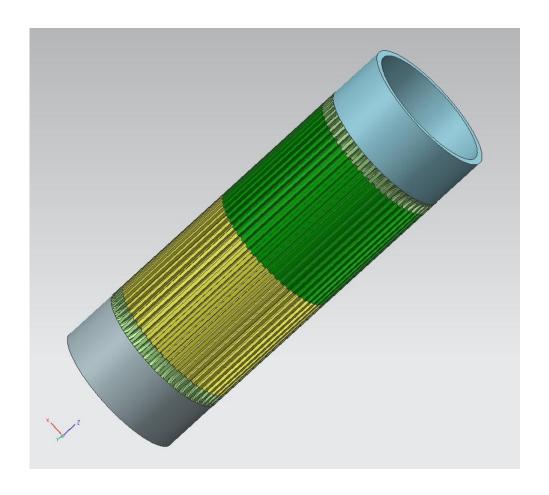
Inside view of furnace chamber



Sintering at high temperature

### **Net Shape Liner Fabrication by FAST**

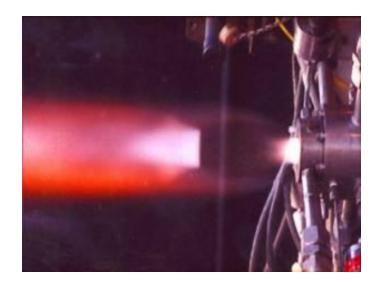




Parts are diffusion bonded to make the liner

### Hot fire Testing of NARloy-Z-Diamond Liner





Combustion chamber liner and test assembly

Hot fire testing at MSFC Test stand

Test data will be analyzed to determine performance